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Research paper Grasshopper Optimisation Algorithm: Theory and application

ABSTRACT

unknown search spaces.

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1. Introduction

The process of finding the best values for the variables of a particular problem to minimise or maximise an objective function is called optimisation. Optimisation problems exist in different fields of studies. To solve an optimisation problem, different steps need to be taken. Firstly, the parameters of the problem should be identified. Based on the nature of the parameters, problems may be classified as continuous or discrete. Secondly, the constraints that are applied to the parameters have to be recognised [1]. Constraints divide the optimisation problems into constrained and unconstrained. Thirdly, the objectives of the given problem should be investigated and considered. In this case, optimisation problems are classified into single-objective versus multi-objective problems [2]. Finally, based on the identified types of parameters, constraints, and number of objectives a suitable optimiser should be chosen and employed to solve the problem.

Mathematical optimisation mainly relies on gradient-based information of the involved functions in order to find the optimal solution. Although such techniques are still being used by different researchers, they have some disadvantages. Mathematical optimisation approaches suffer from local optima entrapment. This refers to an algorithm assuming a local solution is the global solution, thus failing to obtain the global optimum. They are also often ineffective for problems with unknown or computationally expensive

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This paper proposes an optimisation algorithm called Grasshopper Optimisation Algorithm (GOA) and ap-

plies it to challenging problems in structural optimisation. The proposed algorithm mathematically mod-

els and mimics the behaviour of grasshopper swarms in nature for solving optimisation problems. The

GOA algorithm is first benchmarked on a set of test problems including CEC2005 to test and verify its performance qualitatively and quantitatively. It is then employed to find the optimal shape for a 52-bar

truss, 3-bar truss, and cantilever beam to demonstrate its applicability. The results show that the pro-

posed algorithm is able to provide superior results compared to well-known and recent algorithms in the

literature. The results of the real applications also prove the merits of GOA in solving real problems with

derivation [3]. Another type of optimisation algorithm that alleviates these two drawbacks is stochastic optimisation [4].

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Stochastic methods rely on random operators that allow them to avoid local optima. They all start optimisation process by creating one or a set of random solutions for a given problem. In contrast to mathematical optimisation techniques, they do not need to calculate the gradient of a solution, just evaluating the solutions using the objective function(s). Decisions as to how to improve the solutions are made based on the calculated objective values. Therefore, the problem is considered as a black box, which is a very useful mechanism when solving real problems with unknown search spaces. Due to these advantages, stochastic optimisation techniques have become very popular over the past two decades [5].

Among stochastic optimisation approaches, nature-inspired, population-based algorithms are the most popular [6]. Such techniques mimic natural problems-solving methods, often those used by creatures. Survival is the main goal for all creatures. To achieve this goal, they have been evolving and adapting in different ways. Therefore, it is wise to seek inspiration from nature as the best and oldest optimiser on the planet. Such algorithms are classified into two main groups: single-solutions-based and multi-solution-based. In the former class, a single random solution is generated and improved for a particular problem. In the latter class, however, multiple solutions are generated and enhanced for a given problem. Multi-solution-based algorithms are more popular than single-solution-based methods, as the literature shows [7].

Multi-solution-based algorithms intrinsically have higher local optima avoidance due to improving multiple solutions during optimisation. In this case, a trapped solution in a local optimum can be assisted by other solutions to jump out of the local

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optimum. Multiple solutions explore a larger portion of the search space compared to single-solution-based algorithms, so the probability of finding the global optimum is high. Also, information about the search space can be exchanged between multiple solutions, which results in quick movement towards the optimum. Although multi-solution-based algorithms have several advantages, they require more function evaluations.

The most popular single-solution-based algorithms are hill climbing [8] and simulated annealing [9]. Both algorithms follow a similar idea, but the local optima avoidance of SA is higher due to the stochastic cooling factor. Other recent single-solution-based algorithms are Tabu Search (TS) [10,11], and Iterated Local Search (ILS) [12]. The most popular multi-solutions-based algorithms are Genetic Algorithms (GA) [13], Particle Swarm Optimisation (PSO) [14], Ant Colony Optimisation (ACO) [15], and Differential Evolution (DE) [16]. The GA algorithm was inspired by the Darwinian theory of evolution. In this algorithm, solutions are considered as individuals and the parameters of solutions take the place of their genes. Survival of the fittest individuals is the main inspiration of this algorithm where the best individuals tend to participate more in improving poor solutions. The PSO algorithm simulates the foraging of herds of birds or schools of fishes. In this algorithm the solutions are improved with respect to the best solutions obtained so far by each of the particles and the best solution found by the swarm. The ACO algorithm mimics the collective behaviour of ants in finding the shortest path from the nest to the source of foods. Finally, DE utilises simple formulae combining the parameters of existing solutions to improve the population of candidate solutions for a given problem.

The similarity of both classes of nature-inspired algorithms is the improvement of solutions until the satisfaction of an end criterion and the division of optimisation process into two phases: exploration versus exploitation [17]. Exploration refers to the tendency for an algorithm to have highly randomised behaviour so that the solutions are changed significantly. Large changes in the solutions cause greater exploration of the search space and consequently discovery of its promising regions. As an algorithm tends toward exploitation, however, solutions generally face changes on a smaller scale and tend to search locally. A proper balance of exploration and exploitation can result in finding the global optimum of a given optimisation problem.

The literature shows that there are many recent swarm intelligence optimisation techniques such as Dolphin Echolocation (DEL) [18,19], Firefly Algorithm (FA) [20,21], Bat Algorithm (BA) [22], and Grey Wolf Optimizer (GWO) [3]. DEL and BA mimic echolocation of dolphins in finding prey and bats navigating respectively. However, FA simulates the mating behaviour of fireflies in nature. Cuckoo Search (CS) [23,24] is another recent algorithm in this field, in which the reproductive processes of cuckoos are employed to propose an optimisation algorithm. The GWO is also a swarm-based technique that models the hunting mechanism of grey wolves.

There are also other algorithms with different inspiration in the literature. For instance, State of Matter Search (SMS) [25,26] uses the concepts of different phases in matter to optimise problems and the Flower Pollination Algorithm (FPA) [27] has been inspired by the survival and reproduction of flowers using pollination. There is a question here as to why we need more algorithms despite the many algorithms proposed so far.

The answer to this question is in the No Free Lunch (NFL) theorem [28] that logically has proven that there is no optimisation technique for solving all optimisation problems. In other words, algorithms in this field perform equally on average when considering *all* optimisation problems. This theorem, in part, has motivated the rapidly increasing number of algorithms proposed over the last decade and is one of the motivations of this paper as well. The next section proposes a new algorithm mimicking the behaviour of grasshopper swarms. There are a few works in the literature that have tried to simulate locust swarm [29–33]. The current study is an attempt to more comprehensively model grasshopper behaviours and propose an optimisation algorithm based on their social interaction.

Due to their simplicity, gradient-free mechanism, high local optima avoidance, and considering problems as black boxes, natureinspired algorithms have been applied widely in science and industry [34–36]. Therefore, we also investigate the application of the proposed algorithm in solving real problems. The rest of the paper is organised as follows:

The Grasshopper Optimisation Algorithm is proposed in Section 2. Section 3 presents and discusses the results on the optimisation test beds and inspects the behaviour of the proposed algorithm. Section 4 contains the application of the proposed method in the field of structural design optimisation. Finally, Section 5 concludes the work and suggests several directions for future studies.

2. Grasshopper Optimisation Algorithm (GOA)

Grasshopper are insects. They are considered a pest due to their damage to crop production and agriculture. The life cycle of grasshoppers is shown in Fig. 1. Although grasshoppers are usually seen individually in nature, they join in one of the largest swarm of all creatures [37]. The size of the swarm may be of continental scale and a nightmare for farmers. The unique aspect of the grasshopper swarm is that the swarming behaviour is found in both nymph and adulthood [38]. Millions of nymph grasshoppers jump and move like rolling cylinders. In their path, they eat almost all vegetation. After this behaviour, when they become adult, they form a swarm in the air. This is how grasshoppers migrate over large distances.

The main characteristic of the swarm in the larval phase is slow movement and small steps of the grasshoppers. In contrast, longrange and abrupt movement is the essential feature of the swarm in adulthood. Food source seeking is another important characteristic of the swarming of grasshoppers. As discussed in the introduction, nature-inspired algorithms logically divide the search process into two tendencies: exploration and exploitation. In exploration, the search agents are encouraged to move abruptly, while they tend to move locally during exploitation. These two functions, as well as target seeking, are performed by grasshoppers naturally. Therefore, if we find a way to mathematically model this behaviour, we can design a new nature-inspired algorithm.

The mathematical model employed to simulate the swarming behaviour of grasshoppers is presented as follows [39]:

$$X_i = S_i + G_i + A_i \tag{2.1}$$

where X_i defines the position of the *i*-th grasshopper, S_i is the social interaction, G_i is the gravity force on the *i*-th grasshopper, and A_i shows the wind advection. Note that to provide random behaviour the equation can be written as $X_i = r_1 S_i + r_2 G_i + r_3 A_i$ where r_1 , r_2 , and r_3 are random numbers in [0,1].

$$S_{i} = \sum_{\substack{j=1\\j\neq i}}^{N} s(d_{ij}) \widehat{d_{ij}}$$

$$(2.2)$$

where d_{ij} is the distance between the *i*-th and the *j*-th grasshopper, calculated as $d_{ij} = |x_j - x_i|$, *s* is a function to define the strength of social forces, as shown in Eg. (2.3), and $\widehat{d_{ij}} = \frac{x_j - x_i}{d_{ij}}$ is a unit vector from the *i*th grasshopper to the *j*th grasshopper.

The *s* function, which defines the social forces, is calculated as follows:

$$s(r) = f e^{\frac{-r}{l}} - e^{-r}$$
(2.3)



Fig. 1. (a) Real grasshopper (b) Life cycle of grasshoppers (left image courtesy of Mehrdad Momeny).



Fig. 2. (left) Function *s* when l=1.5 and f=0.5 (right) range of function *s* when *x* is in [1,4].



Fig. 3. Behaviour of the function s when varying l or f.

where f indicates the intensity of attraction and l is the attractive length scale.

The function *s* is illustrated in Fig. 2 to show how it impacts on the social interaction (attraction and repulsion) of grasshoppers.

It may be seen in this figure that distances from 0 to 15 are considered. Repulsion occurs in the interval [0 2.079]. When a grasshopper is 2.079 units away from another grasshopper, there is neither attraction nor repulsion. This is called the comfort zone or comfortable distance. Fig. 2 also shows that the attraction increases from 2.079 unit of distance to nearly 4 and then gradually decreases. Changing the parameters l and f in Eq. (2.3) results in different social behaviours in artificial grasshoppers. To see the effects of these two parameters, the function s is re-drawn in Fig. 3 varying l and f independently. This figure shows that the parameters

ters *l* and *f* change comfort zone, attraction region, and repulsion region significantly. It should be noted that the attraction or repulsion regions are very small for some values (l = 1.0 or f = 1.0 for instance). From all these values we have chosen l = 1.5 and f = 0.5.

A conceptual model of the interactions between grasshoppers and the comfort zone using the function s is illustrated in Fig. 4. It may be noted that, in simplified form, this social interaction was the motivating force in some earlier locust swarming models [32].

Although the function s is able to divide the space between two grasshoppers into repulsion region, comfort zone, and attraction region, this function returns values close to zero with distances greater than 10 as Figs. 2 and 3 show. Therefore, this function is not able to apply strong forces between grasshoppers with large distances between them. To resolve this issue, we have mapped



Fig. 4. Primitive corrective patterns between individuals in a swarm of grasshoppers.

the distance of grasshoppers in the interval of [1,4]. The shape of the function *s* in this interval is shown in Fig. 2 (right).

The *G* component in Eq. (2.1) is calculated as follows:

$$G_i = -g\widehat{e_g} \tag{2.4}$$

where *g* is the gravitational constant and \hat{e}_g shows a unity vector towards the centre of earth.

The A component in Eq. (2.1) is calculated as follows:

 $A_i = u\widehat{e_w} \tag{2.5}$

where u is a constant drift and $\widehat{e_w}$ is a unity vector in the direction of wind.

Nymph grasshoppers have no wings, so their movements are highly correlated with wind direction.

Substituting S, G, and A in Eq.(2.1), this equation can be expanded as follows:

$$X_{i} = \sum_{\substack{j=1\\j\neq i}}^{N} s\left(\left|x_{j} - x_{i}\right|\right) \frac{x_{j} - x_{i}}{d_{ij}} - g\widehat{e_{g}} + u\widehat{e_{w}}$$
(2.6)

where $s(r) = fe^{\frac{-r}{T}} - e^{-r}$ and *N* is the number of grasshoppers.

Since nymph grasshoppers land on the ground, their position should not go below a threshold. However, we will not utilise this equation in the swarm simulation and optimisation algorithm because it prevents the algorithm from exploring and exploiting the search space around a solution. In fact, the model utilised for the swarm is in free space. Therefore, Eq. (2.6) is used and can simulate the interaction between grasshoppers in a swarm. The behaviour of two swarms in 2D and 3D space using this equation is illustrated in Figs. 5 and 6. In these two figures, 20 artificial grasshoppers are required to move over 10 units of time.

Fig. 5 shows how Eq. (2.6) brings the initial random population closer until they form a united, regulated swarm. After 10 units of time, all the grasshoppers reach the comfort zone and do not move anymore. The same behaviour is observed in a 3D space in Fig. 6. This shows that the mathematical model is able to simulate a swarm of grasshoppers in 2D, 3D, and hyper dimensional spaces.

However, this mathematical model cannot be used directly to solve optimisation problems, mainly because the grasshoppers quickly reach the comfort zone and the swarm does not converge to a specified point. A modified version of this equation is proposed as follows to solve optimisation problems:

$$X_i^d = c \left(\sum_{\substack{j=1\\j \neq i}}^N c \frac{ub_d - lb_d}{2} s\left(\left| x_j^d - x_i^d \right| \right) \frac{x_j - x_i}{d_{ij}} \right) + \widehat{T}_d$$
(2.7)

where ub_d is the upper bound in the *D*th dimension, lb_d is the lower bound in the *D*th dimension $s(r) = fe^{\frac{-r}{T}} - e^{-r}$, \widehat{T}_d is the value of the *D*th dimension in the target (best solution found so far), and *c* is a decreasing coefficient to shrink the comfort zone, repulsion zone, and attraction zone. Note that *S* is almost similar to the *S* component in Eq. (2.1). However, we do not consider gravity (no *G* component) and assume that the wind direction (*A* component) is always towards a target (\widehat{T}_d).

Eq. (2.7) shows that the next position of a grasshopper is defined based on its current position, the position of the target, and the position of all other grasshoppers. Note that the first component of this equation considers the location of the current grasshopper with respect to other grasshoppers. In fact, we have considered the status of all grasshoppers to define the location of search agents around the target. This is different to PSO as the most well-regarded swarm intelligence technique in the literature. In PSO, there are two vectors for each particle: position and velocity vector. However, there is only one position vector for every search agent in GOA. The other main difference between these two algorithms is that PSO updates the position of particles with respect to current position, personal best, and global best. However, GOA updates the position of a search agent based on its current position, global best, and the position of all other search agents. This means that in PSO none of the other particles contribute to updating the position of a particle, whereas GOA requires all search agents to get involved in defining the next position of each search agent.

It is also worth mentioning here that the adaptive parameter c has been used twice in Eq. (2.7) for the following reasons:

- The first *c* from the left is very similar to the inertial weight (*w*) in PSO. It reduces the movements of grasshoppers around the target. In other words, this parameter balances exploration and exploitation of the entire swarm around the target.
- The second *c* decreases the attraction zone, comfort zone, and repulsion zone between grasshoppers. Considering the component $c\frac{ub_d-ld_d}{2}s(|x_j x_i|)$ in the Eq. (2.7), $c\frac{ub_d-ld_d}{2}$ linearly decreases the space that the grasshoppers should explore and exploit. The component $s(|x_j x_i|)$ indicates if a grasshopper should be repelled from (explore) or attracted to (exploitation) the target.

It should be noted that the inner c contributes to the reduction of repulsion/attraction forces between *grasshoppers* proportional to the number of iterations, while the outer c reduces the search coverage around the *target* as the iteration count increases.

In summary, the first term of Eq. (2.7), the sum, considers the position of other grasshoppers and implements the interaction of grasshoppers in nature. The second term, \hat{T}_d , simulates their tendency to move towards the source of food. Also, the parameter *c* simulates the deceleration of grasshoppers approaching the source of food and eventually consuming it. To provide more random behaviour, and as an alternative, both terms in Eq. (2.7) can be multiplied with random values. Also, individual terms can be multiplied with random values to provide random behaviour in either interaction of grasshoppers or tendency towards the food source.

The proposed mathematical formulations are able to explore and exploit the search space. However, there should be a mechanism to require the search agents to tune the level of exploration to exploitation. In nature, grasshoppers first move and search for foods locally because in larvae they have no wing. They then move freely in air and explore a much larger scale region. In stochastic optimisation algorithms, however, exploration comes first due to the need for finding promising regions of the search space. After finding promising regions, exploitation obliges search agents to



Fig. 6. Behaviour of swarm in a 3D space.

search locally to find an accurate approximation of the global optimum.

For balancing exploration and exploitation, the parameter c is required to be decreased proportional to the number of iteration. This mechanism promotes exploitation as the iteration count increases. The coefficient c reduces the comfort zone proportional to

the number of iterations and is calculated as follows:

$$c = cmax - l\frac{cmax - cmin}{L}$$
(2.8)

where *cmax* is the maximum value, *cmin* is the minimum value, *l* indicates the current iteration, and *L* is the maximum number of iterations. In this work, we use 1 and 0.00001 for *cmax* and *cmin* respectively.



Fig. 7. (a) Behaviour of grasshoppers around a stationary and mobile target in 2D space and (b) 3D space (c) Behaviour of grasshoppers on a unimodal test function and a multi-modal test function.

The effect of this parameter on the movement and convergence of grasshoppers is illustrated in Fig. 7. The sub-figures illustrate the position history of grasshoppers over 100 iterations. We have performed the experiment on both stationary and mobile targets to see how the swarm moves towards and chases them. This figure shows that the swarm converges gradually towards a stationary target in both 2D and 3D spaces. This behaviour is due to reducing the comfort zone by the factor *c*. Fig. 7 also shows that the swarm properly chases a mobile target as well. This is due to the last component of Eq. (2.6) (\hat{T}_d), in which grasshoppers are attracted towards the target. The interesting pattern is the gradual convergence of grasshoppers towards the target over the course of iteration, which is again due to decreasing the factor *c*. These behaviours will assist the GOA algorithm not to converge towards the target too quickly and consequently not to become trapped in local optima. In the last steps of optimisation, however, grasshoppers will converge towards the target as much as possible, which is essential in exploitation.

The above discussions show that the mathematical model proposed requires grasshoppers to move towards a target gradually

Initialize the swarm X_i $(i = 1, 2,, n)$
Initialize cmax, cmin, and maximum number of iterations
Calculate the fitness of each search agent
$T=the \ best \ search \ agent$
while $(l < Max number of iterations)$
Update c using Eq. (2.8)
for each search agent
Normalize the distances between grasshoppers in [1,4]
Update the position of the current search agent by the equation (2.7)
Bring the current search agent back if it goes outside the boundaries
end for
Update T if there is a better solution
l=l+1
end while
Return T

Fig. 8. Pseudo codes of the GOA algorithm.

over the course of iterations. In a real search space, however, there is no target because we do not know exactly where the global optimum, the main target, is. Therefore, we have to find a target for grasshoppers in each step of optimisation. In GOA, it is assumed that the fittest grasshopper (the one with the best objective value) during optimisation is the target. This will assist GOA to save the most promising target in the search space in each iteration and requires grasshoppers to move towards it. This is done with the hope of finding a better and more accurate target as the best approximation for the real global optimum in the search space. Fig. 7 includes two test functions and shows that the proposed model and target updating mechanism are effective in problems with unknown optimum as well.

The pseudo code of the GOA algorithm is shown in Fig. 8. The GOA starts optimisation by creating a set of random solutions. The search agents update their positions based on Eq. (2.7). The position of the best target obtained so far is updated in each iteration. In addition, the factor c is calculated using Eq. (2.8) and the distances between grasshoppers are normalised in [1,4] in each iteration. Position updating is performed iteratively until the satisfaction of an end criterion. The position and fitness of the best target is finally returned as the best approximation for the global optimum.

Although the above simulations and discussions demonstrate the effectiveness of the GOA algorithm in finding the global optimum in a search space, the performance of the proposed algorithm is investigated by employing a set of mathematical functions and three challenging real problems in the next sections. Note that the source codes of the GOA algorithm can be found at http://www.alimirjalili.com/Projects.html and http://au.mathworks. com/matlabcentral/profile/authors/2943818-seyedali-mirjalili.

3. Results

This section first presents the test bed problems and performance metrics that are used to benchmark the performance of the proposed GOA algorithm. The experimental results are then provided and analysed in detail.

3.1. Experimental setup

In the field of stochastic optimisation, it is common to employ a set of mathematical test functions with known optima. Thus, the performance of different algorithms can be measured quantitatively. However, the characteristics of the test functions should be diverse to be able to draw a mature conclusion. In this work, three sets of test functions with different features are employed to confidently benchmark the performance of the proposed algorithm. The test functions are unimodal, multimodal, and composite [40–43]. The mathematical formulation of these test functions are available in the appendix. As shown in Fig. 9, a unimodal test function has no local solutions and there is only one global optimum. The entire search space favours the global optima, so the convergence speed and exploitation of an algorithm can be benchmarked. Fig. 9 also shows that multi-modal and composite test functions have many local optima which make them highly suitable for benchmarking the performance of an algorithm in terms of local optima avoidance and exploration. Composite tests functions are usually more challenging than the multi-modal test functions and better mimic real search spaces. Therefore, the potential performance of an algorithm solving real problems may be inferred from such benchmarks.

For solving the test functions, 30 search agents and 500 iterations were employed. Each of the test functions was solved 30 times to generate the statistical results. Different performance indicators were utilised to quantitatively compare the algorithms: average and standard deviation of the best solutions obtained in the last iterations. Obviously, the lower the value of average and standard deviation, the greater the ability of an algorithm in avoiding local solutions and determining the global optimum. Qualitative results, including convergence curves, trajectory of grasshoppers, search history, and average fitness of population have been illustrated and analysed in the following subsection.

For verification of results, seven algorithms were employed from the literature including well-known and recent ones: PSO, SMS [25,26], BA [22], FPA [27], CS [23,24], FA [20,21], GA, DE, and Gravitational Search Algorithm (GSA) [44]. The initial controlling parameters of all algorithms are shown in Table 1.

3.2. Qualitative results and discussion

The first experiment was performed on the 2D version of some of the test functions using only 5 artificial grasshoppers. The main objective for this experiment was to observe the behaviour of the GOA qualitatively. Five diagrams have been drawn for each of the test functions in Fig. 10 in addition to the shape of test functions. These diagrams are:

- Search history: this diagram shows the location history of the artificial grasshoppers during optimisation.
- Attraction/repulsion rates: this diagram shows the number of times that all artificial grasshoppers attracted or repelled each other during optimisation.
- Trajectory of the first grasshopper in the first dimension: this diagram shows the value of the first variable of the first grasshopper in each iteration.
- Average fitness: this diagram indicates the average objective value of all grasshoppers in each iteration.
- Convergence curve: this diagram shows the objective value of the best solutions obtained so far (target) in each iteration.

As per the results in Fig. 10, grasshoppers tend to explore the promising regions of the search space and cluster around the global optima eventually. This pattern can be observed in unimodal, multimodal, and composite test functions. These results show that the GOA algorithm beneficially balances exploration and exploitation to drive the grasshoppers towards the global optimum. The rates of attraction and repulsion in the pie charts show that the grasshoppers interact differently on the test functions. They seem to attract each other more often on the unimodal test functions. This can clearly be observed in the pie charts for F1 and F4. This behaviour is fruitful because unimodal functions do not have local optima, so grasshopper can better determine the global optimum by moving towards the best solution obtained so far.

Another interesting pattern in the pie charts is the high repulsion rate between grasshoppers when solving multi-modal and composite test functions. This can be observed in the pie charts



Fig. 9. An example of unimodal, multi-modal, and composite test functions.

Table 1

Initial values for the controlling parameters of algorithms.

Algorithm	Parameter	Value
PSO	Topology Cognitive and social constants Inertial weight	Fully connected 1.5, 1.5 Linearly decreases from 0.6 to 0.3
GA	Type Selection Crossover Mutation	Real coded Roulette wheel Single point (probability = 1) Uniform (probability = 0.01)
DE	Crossover probability Differential weight	0.9 0.5
GSA	Rnorm, Rpower, alpha, and G_0	2, 1, 20, 100
BA	Loudness (A), pulse rate (r) Frequency min and max	0.5, 0.5 0, 2
FPA	probability switch(p)	0.4
SMS	Beta Alpha H Phases	[0.9, 0.5, 0.1] [0.3, 0.05, 0] [0.9, 0.2, 0] [0.5, 0.1,-0.1]
FA	Alpha, beta, and gamma	0.2, 1, 1

of F9, F14 and F18. This is due to the fact that repulsion is a key mechanism to avoid local solutions in the GOA algorithm and these results show that this algorithm prevents grasshoppers from local optima stagnation by high repulsion rates. It is worth mentioning here that the results show that the high repulsion rate does not negatively impact on the convergence. This is due to the adaptive parameter of GOA, which shrinks the repulsion area proportional to the number of iterations. Therefore, grasshoppers avoid local valleys in the initial steps of iteration and cluster around the global optimum in the final stages of optimisation. For the test functions with both unimodal and multi modal regions (F10 for instance), Fig. 10 shows that the repulsion rate is lower. These results again demonstrate that GOA efficiently balances exploration and exploitation to approximate the global optimum.

The trajectory curves in Fig. 10 show that the grasshoppers exhibit large, abrupt changes in the initial steps of optimisation. This is due to the high repulsion rate which causes exploration of the search space by GOA. It also can be seen that the fluctuation decreased gradually during optimisation, which is due to the adaptive comfort zone and attraction forces between the grasshoppers. This guarantees that the proposed GOA algorithm explores and exploits the search space and converges towards a point eventually. To confirm that this behaviour results in improving the fitness of grasshoppers, average fitness of grasshoppers and convergence curves are provided in Fig. 10. The curves clearly show descending behaviour on all of the test functions. This proves that GOA enhances the initial random population on the test functions and desirably improves the accuracy of the approximated optimum over the course of iterations.

3.3. Quantitative results and discussion

The above discussed results qualitatively demonstrated that the GOA is able to solve optimisation problems. However, the test functions were of 2 variables and qualitative results cannot tell us how much better this algorithm is compared to current ones. In order to show the merits of GOA quantitatively, this subsection solved the test functions with 30 dimensions and presents/discusses the results quantitatively. The experimental results are provided in Tables 2, 3, and 4 for unimodal, multi-modal, and composite test functions. Note that the results are normalised between 0 and 1 for all the test functions due to the different domain/range of test functions. This assist us in conveniently comparing the results on different test functions as well.

As per the results in Table 2, the GOA algorithm shows the best results when solving unimodal test functions. The results of this algorithm are substantially better in more than half of the unimodal test functions, showing the high performance of this algorithm. Unimodal test functions have only one global optimum, so the results clearly show that the GOA algorithm benefits from high exploitation ability.

The results in Table 3 are consistent with those in Table 2, in which the GOA algorithm tends to significantly outperform others in both of the performance metrics. The results of this algorithm are again remarkably superior in the majority of multi-modal test functions. Since the multi-modal test functions have a significant number of local solutions, these results quantitatively show the effectiveness of the proposed algorithm in avoiding local solutions during optimisation.

The results of the algorithms on composite test functions are presented in Table 4. These results show that the GOA algorithm provides very competitive results compared to other algorithms. Composite test functions are even more challenging than the multi-modal ones and require a proper balance between exploration and exploitation. Therefore, it can be stated that the GOA is able to balance exploration and exploitation properly for solving such challenging problems.

Comparing algorithms based on average and standard deviation over 30 independent runs does not compare each of the runs. Therefore, it is still possible that the superiority occurs by chance despite its low probability in 30 runs. In order to compare the results of each run and decide on the significance of the results, the Wilcoxon statistical test was performed at 5% significance level and the p-values are reported in Table 5. For the statistical test, the best algorithm in each test function is chosen and compared with other algorithms independently. For example, if the best algorithm is GOA, a pairwise comparison is done between GOA/PSO, GOA/GSA, GOA/BA, and so on. Note that since the best algorithm cannot be compared with itself, N/A has been



Fig. 10. Behaviour of GOA on the 2D benchmark problems.

written for the best algorithm in each function which stands for Not Applicable.

As per the results in this table, p-values are mostly less than 0.05 for the GOA, which demonstrates that the superiority of this algorithm is statistically significant. For the F3 function the results

show the FPA is not significantly superior to GOA. Overall, these results show that GOA is able to outperform other algorithms in the literature. According to the NFL theorem, therefore, it has the potential to solve problems (of the types tested) that cannot be solved efficiently by other algorithms.

Table 2	
Results of unimodal	benchmark functions.

F	GOA		PSO	PSO		GSA		BA	
	Ave	std	ave	std	ave	std	ave	std	
F1	0.0000	0.0000	0.2391	0.5622	0.0002	0.0012	0.9882	1.0000	
F2	0.0020	0.0010	0.0097	0.0013	0.0000	0.0000	1.0000	1.0000	
F3	0.0010	0.0203	0.2613	0.3547	0.0328	0.0395	1.0000	1.0000	
F4	0.0000	0.0000	0.4767	0.4730	0.3244	0.5119	0.9148	1.0000	
F5	0.0000	0.0000	0.0386	0.0944	0.0000	0.0000	1.0000	1.0000	
F6	0.0000	0.0000	0.7786	0.4808	0.3825	0.2231	1.0000	1.0000	
F7	0.0000	0.0000	0.1349	0.1648	0.0226	0.0763	1.0000	1.0000	
	FPA		SMS						
F	FPA		SMS		FA		GA		
F	FPA ave	std	SMS ave	std	FA ave	std	GA ave	std	
F 	FPA <i>ave</i> 0.0329	std 0.0784	SMS <i>ave</i> 1.0000	std 0.4478	FA ave 0.1581	std 0.0748	GA <i>ave</i> 0.4121	std 0.5202	
F F1 F2	FPA ave 0.0329 0.0131	std 0.0784 0.0007	SMS ave 1.0000 0.0157	std 0.4478 0.0008	FA ave 0.1581 0.0076	std 0.0748 0.0001	GA <i>ave</i> 0.4121 0.0100	std 0.5202 0.0003	
F F1 F2 F3	FPA ave 0.0329 0.0131 0.0000	std 0.0784 0.0007 0.0000	SMS ave 1.0000 0.0157 0.3486	<i>std</i> 0.4478 0.0008 0.0651	FA ave 0.1581 0.0076 0.0617	<i>std</i> 0.0748 0.0001 0.0160	GA ave 0.4121 0.0100 0.2022	std 0.5202 0.0003 0.0710	
F F1 F2 F3 F4	FPA ave 0.0329 0.0131 0.0000 0.3219	std 0.0784 0.0007 0.0000 0.4215	SMS ave 1.0000 0.0157 0.3486 1.0000	<i>std</i> 0.4478 0.0008 0.0651 0.7232	FA ave 0.1581 0.0076 0.0617 0.3796	std 0.0748 0.0001 0.0160 0.2116	GA ave 0.4121 0.0100 0.2022 0.7245	std 0.5202 0.0003 0.0710 0.2384	
F F1 F2 F3 F4 F5	FPA ave 0.0329 0.0131 0.0000 0.3219 0.0060	<i>std</i> 0.0784 0.0007 0.0000 0.4215 0.0345	SMS ave 1.0000 0.0157 0.3486 1.0000 0.3901	std 0.4478 0.0008 0.0651 0.7232 0.4781	FA ave 0.1581 0.0076 0.0617 0.3796 0.0068	<i>std</i> 0.0748 0.0001 0.0160 0.2116 0.0068	GA ave 0.4121 0.0100 0.2022 0.7245 0.0746	<i>std</i> 0.5202 0.0003 0.0710 0.2384 0.0931	
F F1 F2 F3 F4 F5 F6	FPA ave 0.0329 0.0131 0.0000 0.3219 0.0060 0.0088	std 0.0784 0.0007 0.0000 0.4215 0.0345 0.0189	SMS ave 1.0000 0.0157 0.3486 1.0000 0.3901 0.7025	std 0.4478 0.0008 0.0651 0.7232 0.4781 0.5891	FA ave 0.1581 0.0076 0.00617 0.3796 0.0068 0.0414	std 0.0748 0.0001 0.0160 0.2116 0.0068 0.0190	GA ave 0.4121 0.0100 0.2022 0.7245 0.0746 0.1933	std 0.5202 0.0003 0.0710 0.2384 0.0931 0.1932	

Table 3

Results of multimodal benchmark functions.

F	GOA		PSO	PSO		GSA			
	ave	std	ave	std	ave	std	ave	std	
F8	1.0000	0.0002	0.7425	0.0016	0.8473	0.0020	0.0148	1.0000	
F9	0.0000	0.0007	0.6520	1.0000	0.1361	0.2722	0.7022	0.7517	
F10	0.0975	1.0000	0.6140	0.2426	0.0000	0.0000	0.9665	0.1155	
F11	0.0000	0.0000	0.8184	0.3512	1.0000	0.5790	0.9912	1.0000	
F12	0.0000	0.0007	0.4689	0.8147	0.0577	0.4246	0.6892	0.9635	
F13	0.0000	0.0000	0.0973	0.1647	0.1603	0.0890	1.0000	1.0000	
F	FPA		SMS	SMS		FA		GA	
	ave	std	ave	std	ave	std	ave	std	
F8	0.0381	0.0006	0.5613	0.0049	0.6140	0.0000	0.0000	0.0006	
F9	0.6568	0.4179	0.8628	0.2633	0.8377	0.0329	1.0000	0.0000	
F10	0.7170	0.3848	1.0000	0.0666	0.7078	0.0410	0.8628	0.1415	
F11	0.0124	0.0058	0.6746	0.7789	0.0548	0.0070	0.1941	0.2865	
F12	0.0237	0.1907	0.1140	0.0000	0.2442	1.0000	1.0000	0.4363	
F13	0.3766	0.1566	0.9609	0.2394	0.1119	0.1510	0.4446	0.0798	

Table 4

Results of composite benchmark functions.

F	GOA		PSO	PSO		GSA		BA	
	Ave	std	ave	std	ave	std	ave	std	
F14	0.0000	0.3386	0.6083	1.0000	0.0840	0.2977	1.0000	0.5714	
F15 F16	0.4892	0.7182	0.4236	0.7929 0.6805	0.0672	0.5226	1.0000	1.0000	
F17	0.8169	1.0000	0.3241	0.2970	0.0000	0.6439	1.0000	0.6905	
F18 F19	0.0000 0.7863	0.0064 0.9355	0.3122 1.0000	0.0000	0.0581 0.9391	0.2503	1.0000 0.9097	0.8953 0.4190	
F	FPA		SMS		FA		GA		
	ave	std	ave	std	ave	std	ave	std	
F14	0.0008	0.0570	0.5604	0.4830	0.5035	0.6008	0.3799	0.0000	
F15	0.0000	0.4822	0.5097	0.5559	0.5730	0.9765	0.1338	0.0000	
F16	0.3381	0.0759	0.8914	0.5077	0.4921	0.0922	0.6820	0.1783	
F17	0.1395	0.0348	0.6594	0.0383	0.3264	0.4911	0.3660	0.0000	
F18	0.3249	0.9194	0.3144	0.4097	0.3160	0.3885	0.1347	0.0000	
F19	0.0000	0.0702	0.4257	0.8595	0.7068	1.0000	0.0211	0.1720	

To further show the effectiveness of the proposed GOA algorithm, we have solved more challenging test functions and compared the results with the most popular algorithms in the literature. The test functions are 25 taken from the CEC2005 special session [45]. These test functions are the most challenging test functions in the literature and can be found in the appendix. The results are compared to PSO, GA, DE, GSA, BA, FPA, and FA as the most well-known and recent algorithms in the literature. The results are again normalised in [0,1] and presented in Tables 6 and 7.

Inspecting the results in Table 6, it is evident that the proposed GOA algorithm outperforms other algorithms on the majority of the CEC2005 test functions. The p-values in Table 7 show that the superiority of GOA is statistically significant. Comparison

Table 5						
P-values	obtained	from	the	Wilcoxon	ranksum	test.

ТР	GOA	PSO	GSA	BA	FPA	SMS	FA	GA
F1	N/A	0.000183	N/A	0.000183	0.000183	0.000183	0.000183	0.000183
F2	0.002827	0.000183	N/A	0.000183	0.000183	0.000183	0.000183	0.000183
F3	0.140465	0.000183	0.000183	0.000183	N/A	0.000183	0.000183	0.000183
F4	N/A	0.000183	0.000183	0.000183	0.000183	0.000183	0.000183	0.000183
F5	N/A	0.000183	0.241322	0.000183	0.000183	0.000183	0.000183	0.000183
F6	N/A	0.000183	0.000183	0.000183	0.000183	0.000183	0.000183	0.000183
F7	N/A	0.000183	0.00033	0.000183	0.000183	0.000583	0.000183	0.000183
F8	0.000183	0.000183	0.000183	N/A	0.000183	0.000183	0.000183	0.000183
F9	N/A	0.000183	0.01133	0.000183	0.000183	0.000183	0.000183	0.000183
F10	0.000183	0.000183	N/A	0.000183	0.000183	0.000183	0.000183	0.000183
F11	N/A	0.000183	0.000183	0.000183	0.001315	0.000183	0.000183	0.000183
F12	N/A	0.000183	0.000183	0.000183	0.000183	6.39e-5	0.000183	0.000183
F13	N/A	0.000183	0.000183	0.000183	0.000183	6.39e-5	0.000183	0.000183
F14	N/A	0.001315	0.009108	0.000246	0.025748	0.001706	0.001315	0.002827
F15	0.001008	0.004586	0.791337	0.000246	N/A	0.001008	0.000769	0.002827
F16	N/A	0.000246	0.00044	0.000183	0.000183	0.000183	0.000183	0.000183
F17	0.001008	0.002827	N/A	0.001008	0.002827	0.002827	0.002202	0.002827
F18	N/A	0.01133	0.472676	0.000183	0.025748	0.000183	0.000183	0.000183
F19	0.025748	0.000183	0.000183	0.000183	N/A	0.00033	0.00044	0.021134

with some of the algorithms provide p-values greater than 0.05 occasionally. This shows that the GOA algorithm is not significantly better on those functions. Also, GOA provides very competitive results on the F12_CEC2005, F16_CEC2005, and F24_CEC2005 test functions, since the high p-values show GSA was not significantly better. Since CEC test functions are very challenging and mimic different difficulties of a real search space, these results strongly demonstrate the merits of the proposed GOA algorithm compared to other algorithms in the literature.

To sum up, the discussions and findings of this section clearly demonstrate the quality of exploration, local optima avoidance, exploitation, and convergence of the GOA algorithm. The high exploration and local optima avoidance of this algorithm originates from the high repulsion rate between grasshoppers. The repulsive force requires grasshoppers to avoid each other and explore the search space extensively. This is the main reason for high local optima avoidance of GOA as well. Exploitation and convergence are encouraged by the attraction forces between the grasshoppers, and the adaptive comfort zone. High attractive forces between grasshoppers drive them quickly towards the best solution obtained so far. The adaptive comfort zone coefficient decreases proportional to the number of iterations, generating smaller repulsion forces and emphasising exploitation. The adaptive behaviour of the comfort zone coefficient also results in a proper balance between exploration and exploitation.

Although these findings strongly suggest that GOA is able to solve real problems, in the following section we use three real problems in the field of structural design to demonstrate and confirm the applicability of this algorithm in solving real problems with unknown search spaces.

4. Real applications

Solving structural design problems using stochastic optimisation techniques has been a popular research direction in the literature [46–54]. This section solves three of the conventional structural design problems using the proposed GOA algorithm.

4.1. Three-bar truss design problem

This structural design problem is one of the most widely-used case studies in the literature [55,56]. This problem is formulated as follows:

Consider
$$\vec{x} = [x_1 \ x_2 \] = [A_1 A_2],$$

$$\begin{array}{ll} \text{Minimise} & f(\vec{x}) = \left(2\sqrt{2}x_1 + x_2\right) * l, \\ \text{Subject to} & g_1(\vec{x}) = \frac{\sqrt{2}x_1 + x_2}{\sqrt{2}x_1^2 + 2x_1x_2} P - \sigma \leq 0, \\ & g_2(\vec{x}) = \frac{x_2}{\sqrt{2}x_1^2 + 2x_1x_2} P - \sigma \leq 0, \\ & g_3(\vec{x}) = \frac{1}{\sqrt{2}x_2 + x_1} P - \sigma \leq 0, \end{array}$$

Variable range $0 \le x_1, x_2 \le 1$,

where l = 100 cm, $P = 2 \text{ KN/cm}^2$, $\sigma = 2 \text{ KN/cm}^2$

Fig. 11 shows the shape of this truss and the forces applied. As this figure and the problem formulation show, there are two structural parameters: the area of bars 1 and 3 and area of bar 2. The objective is to minimise the weight of the truss. This problem is subject to several constraints as well: stress, deflection, and buckling constraints.

The proposed GOA with 20 search agents and 650 iterations was employed on this problem. Since this problem is a constrained problem, a constraint handling method needed to be integrated with GOA. For the sake of simplicity, a death penalty has been utilised. It penalises the search agents that violate any of the constraints at any level with a large objective value. For verification, the results are compared to ALO, DEDS, PSO-DE, MBA, Ray and Sain, and Tsa methods and presented in Table 8. This table shows the optimal values for both variables and weight.

Inspecting the results of algorithms on this problem, it is evident that GOA managed to show very competitive results compared to ALO, DEDS, PSO-DE, and MBA with a better maximum function evaluation. Also, this algorithm outperforms the rest of the algorithms significantly. These results show that the GOA algorithm is able to handle the difficulties of a constrained search space efficiently.

4.2. Cantilever beam design problem

This is another popular structural design problem in the literature formulated as follows:

Consider	$\vec{\mathbf{x}} = [\mathbf{x}_1 \ \mathbf{x}_2 \ \mathbf{x}_3 \ \mathbf{x}_4 \ \mathbf{x}_5]$
Minimise	$f(\vec{x}) = 0.6224(x_1 + x_2 + x_3 + x_4 + x_5),$
Subject to	$g(\vec{x}) = \frac{61}{x_1^3} + \frac{27}{x_2^3} + \frac{19}{x_3^3} + \frac{7}{x_4^3} + \frac{1}{x_5^3} - 1 \le 0,$

Table 6				
Results	on	CEC	benchmark	functions.

F	GOA		PSO	PSO			DE	
	Ave	std	ave	std	ave	std	ave	std
F1_CEC2005	0.0000	0.0000	0.6040	1.0000	0.4972	0.8825	0.0424	0.0950
F2_CEC2005	0.0000	0.0000	0.3341	0.1579	0.4860	0.1481	0.3483	0.1134
F3_CEC2005	0.0037	0.0057	0.2299	0.2852	0.3130	0.0441	0.1311	0.0801
F4_CEC2005	0.0000	0.0127	0.4569	0.3523	0.4807	0.0958	0.3726	0.0473
F5_CEC2005	0.0000	0.0602	0.7003	0.4740	0.4788	0.0000	0.2737	0.0513
F6_CEC2005	0.0000	0.0000	0.3078	0.1878	0.2109	0.2577	0.0037	0.0016
F7_CEC2005	0.0000	0.0000	0.8524	0.6681	0.3416	0.3029	0.0759	0.0806
F8_CEC2005	0.5505	0.7277	1.0000	0.0000	0.9702	0.1651	0.9816	0.4399
F9_CEC2005	0.0000	0.0356	0.5743	0.8935	1.0000	0.3927	0.6055	0.1518
F10_CEC2005	0.0000	0.0000	0.5391	1.0000	0.7874	0.0628	0.4904	0.1780
F11_CEC2005	0.7570	1.0000	0.6283	0.8342	0.9965	0.3326	1.0000	0.0000
F12_CEC2005	0.1087	0.3460	0.3531	1.0000	1.0000	0.4454	0.6698	0.2908
F13_CEC2005	0.0000	0.0046	0.1030	0.0580	0.4551	0.0808	0.1484	0.0091
F14_CEC2005	0.0154	0.0855	0.0000	0.9627	0.6053	0.2766	0.5800	0.0083
F15_CEC2005	0.3527	1.0000	0.5065	0.8562	0.6289	0.2734	0.4217	0.3437
F16_CEC2005	0.1520	0.8379	0.3316	0.7594	0.3718	0.0825	0.1878	0.0000
F17_CEC2005	0.0000	0.7666	0.4744	1.0000	0.2712	0.0249	0.1401	0.0000
F18_CEC2005	0.0000	0.0000	0.3981	0.7095	0.5096	0.2305	0.2369	0.2265
F19_CEC2005	0.0000	0.0000	0.3494	0.3857	0.4993	0.2066	0.2122	0.0501
F20 CEC2005	0.0000	0.0000	0.4166	0.8314	0.6018	0.3811	0.2662	0.0663
F21 CEC2005	0.0000	0.0416	0.7469	0.0314	0.8364	0.0467	0.4840	0.3295
F22 CEC2005	0.0000	0.0000	0.4000	0.7118	0.6116	0.2260	0.3587	0.1408
F23 CEC2005	1.0000	0.3113	0.0000	0.6942	0.0608	0.7502	0.2773	0.4302
F24 CEC2005	1.0000	0.0724	0.0000	0.7804	0.0458	0.7228	0.1298	0.2603
F25_CEC2005	0.3449	0.0075	0.8530	0.0765	0.8299	0.0035	0.4108	0.0442
F	GSA		BA		FPA		FA	
F	GSA ave	std	BA ave	std	FPA ave	std	FA ave	std
F F1_CEC2005	GSA <i>ave</i> 0.4318	std 0.5622	BA ave 1.0000	std 0.8805	FPA <i>ave</i> 0.1184	std 0.3697	FA <i>ave</i> 0.1076	std 0.1120
F F1_CEC2005 F2_CEC2005	GSA ave 0.4318 0.2614	std 0.5622 0.0257	BA ave 1.0000 1.0000	std 0.8805 1.0000	FPA ave 0.1184 0.0615	std 0.3697 0.0320	FA ave 0.1076 0.1705	std 0.1120 0.0191
F F1_CEC2005 F2_CEC2005 F3_CEC2005	GSA ave 0.4318 0.2614 0.1419	<i>std</i> 0.5622 0.0257 0.1081	BA ave 1.0000 1.0000 1.0000	<i>std</i> 0.8805 1.0000 1.0000	FPA ave 0.1184 0.0615 0.0000	<i>std</i> 0.3697 0.0320 0.0000	FA ave 0.1076 0.1705 0.0620	<i>std</i> 0.1120 0.0191 0.0640
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724	<i>std</i> 0.5622 0.0257 0.1081 0.4499	BA ave 1.0000 1.0000 1.0000 1.0000	<i>std</i> 0.8805 1.0000 1.0000 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651	std 0.3697 0.0320 0.0000 0.0685	FA ave 0.1076 0.1705 0.0620 0.0661	<i>std</i> 0.1120 0.0191 0.0640 0.0000
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F5_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225	std 0.5622 0.0257 0.1081 0.4499 0.0534	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707	std 0.3697 0.0320 0.0000 0.0685 0.3970	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041	std 0.1120 0.0191 0.0640 0.0000 0.0675
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F5_CEC2005 F6_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104	<i>std</i> 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F5_CEC2005 F6_CEC2005 F7_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F6_CEC2005 F7_CEC2005 F8_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108 0.2817	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 1.0000 0.1382	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476 0.1094
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F5_CEC2005 F7_CEC2005 F8_CEC2005 F8_CEC2005 F9_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013	std 0.8805 1.0000 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.7155	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476 0.1094 0.0000
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F5_CEC2005 F7_CEC2005 F8_CEC2005 F9_CEC2005 F10_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0534 0.0762 0.5975 1.0000 1 0.7690	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.10000 0.1382 0.3595 0.8251	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.7155 0.4252	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0476 0.0476 0.1094 0.0000 0.0945
F F1_CEC2005 F3_CEC2005 F4_CEC2005 F5_CEC2005 F5_CEC2005 F7_CEC2005 F9_CEC2005 F9_CEC2005 F10_CEC2005 F11_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638	std 0.8805 1.0000 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.7155 0.4252 0.9867	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476 0.1094 0.0000 0.0945 0.0856
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F5_CEC2005 F5_CEC2005 F6_CEC2005 F8_CEC2005 F8_CEC2005 F10_CEC2005 F11_CEC2005 F12_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.0000	<i>std</i> 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.1006	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.7155 0.4252 0.9867 0.4274	<i>std</i> 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476 0.1094 0.00945 0.0856 0.4266
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F5_CEC2005 F6_CEC2005 F6_CEC2005 F8_CEC2005 F8_CEC2005 F9_CEC2005 F10_CEC2005 F11_CEC2005 F13_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.0000 0.0000 0.0516	<i>std</i> 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.1006 0.0331	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.7155 0.4252 0.9867 0.4474 0.1328	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476 0.1094 0.0000 0.0945 0.0856 0.4266 0.0061
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F5_CEC2005 F6_CEC2005 F7_CEC2005 F8_CEC2005 F9_CEC2005 F10_CEC2005 F10_CEC2005 F11_CEC2005 F12_CEC2005 F14_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.0000 0.0516 1.0000	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.00331 0.1088	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000 0.7244	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000 0.0000	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.7155 0.4252 0.9867 0.4252 0.9867 0.4474 0.1328 0.4933	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476 0.1094 0.0000 0.0945 0.0856 0.4266 0.0061 0.0425
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F6_CEC2005 F7_CEC2005 F7_CEC2005 F9_CEC2005 F10_CEC2005 F10_CEC2005 F11_CEC2005 F13_CEC2005 F14_CEC2005 F15_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8724 0.8725 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.0000 0.0516 1.0000 0.0000	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.1006 0.0331 0.1088 0.0000	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000 0.7244 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000 0.8301	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733 0.3127	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000 0.0000 0.1340	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.7155 0.4252 0.9867 0.4474 0.1328 0.4933 0.4936	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476 0.1094 0.0000 0.0945 0.0856 0.4266 0.0061 0.0425 0.8700
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F6_CEC2005 F7_CEC2005 F9_CEC2005 F9_CEC2005 F10_CEC2005 F11_CEC2005 F11_CEC2005 F13_CEC2005 F14_CEC2005 F14_CEC2005 F16_CEC2005 F16_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.0516 1.0000 0.0000 0.0000	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.1006 0.0331 0.1088 0.0000 1.0000	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000 0.7244 1.0000 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.0382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000 1.0000 0.8301 0.9453	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733 0.3127 0.1872	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000 0.0000 0.1340 0.1099	FA ave 0.1076 0.1705 0.0620 0.0661 0.10141 0.0125 0.9866 0.7155 0.4252 0.9867 0.4258 0.4256 0.4256 0.4256 0.4364	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476 0.1094 0.0000 0.0945 0.0856 0.4266 0.0061 0.0425 0.8700 0.0588
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F5_CEC2005 F6_CEC2005 F8_CEC2005 F9_CEC2005 F10_CEC2005 F11_CEC2005 F12_CEC2005 F13_CEC2005 F14_CEC2005 F14_CEC2005 F16_CEC2005 F16_CEC2005 F17_CEC2005 F17_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0005	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.1006 0.0331 0.1088 0.0000 1.0000 0.5209	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000 0.7244 1.0000 1.0000 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000 0.6132 1.0000 1.0000 0.8301 0.9453 0.6007	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733 0.3127 0.1872 0.0966	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000 0.0000 0.0000 0.1340 0.1099 0.1335	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.7155 0.4252 0.9867 0.4252 0.9867 0.4474 0.1328 0.4933 0.4956 0.1364 0.1530	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0476 0.0476 0.1094 0.0000 0.0945 0.0856 0.4266 0.0061 0.0425 0.8700 0.0588 0.4022
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F5_CEC2005 F6_CEC2005 F8_CEC2005 F8_CEC2005 F10_CEC2005 F11_CEC2005 F12_CEC2005 F13_CEC2005 F14_CEC2005 F16_CEC2005 F16_CEC2005 F17_CEC2005 F18_CEC2005 F18_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.0516 1.0000 0.0000 0.0001 0.0000 0.0000 0.00055 0.0306	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.0331 0.1088 0.0000 1.0000 0.5209 0.0997	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000 0.7244 1.0000 1.0000 1.0000 1.0000	<i>std</i> 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000 0.8301 0.9453 0.6007 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733 0.3127 0.1872 0.0966 0.1485	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000 0.0000 0.0000 0.0000 0.1340 0.1335 0.4907	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.7155 0.4252 0.9867 0.4252 0.9867 0.4474 0.1328 0.4933 0.4956 0.1364 0.1330 0.1162	<i>std</i> 0.1120 0.0191 0.0640 0.0000 0.0675 0.0476 0.1094 0.0945 0.0945 0.0856 0.4266 0.0061 0.0425 0.8700 0.0588 0.4022 0.0744
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F5_CEC2005 F6_CEC2005 F6_CEC2005 F8_CEC2005 F9_CEC2005 F10_CEC2005 F11_CEC2005 F13_CEC2005 F14_CEC2005 F14_CEC2005 F15_CEC2005 F16_CEC2005 F18_CEC2005 F18_CEC2005 F18_CEC2005 F19_CEC2005 F19_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.0000 0.0000 0.00016 1.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0005 0.0306 0.023	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.1006 0.0331 0.1088 0.0000 1.0000 0.5209 0.0997 0.0478	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000 0.7244 1.0000 1.0000 1.0000 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000 0.8301 0.9453 0.6007 1.0000 1.0000 1.0000 1.0000 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733 0.3127 0.3127 0.1872 0.0966 0.1485 0.1259	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000 0.0000 0.0000 0.1340 0.1099 0.1335 0.4907 0.1073	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0725 0.9866 0.7155 0.4252 0.9866 0.4474 0.1328 0.4933 0.4956 0.1520 0.1621 0.0162 0.0975	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0476 0.1094 0.0000 0.0945 0.0856 0.4266 0.0061 0.0425 0.8700 0.0588 0.4022 0.0744 0.0111
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F6_CEC2005 F7_CEC2005 F9_CEC2005 F9_CEC2005 F10_CEC2005 F11_CEC2005 F12_CEC2005 F14_CEC2005 F14_CEC2005 F15_CEC2005 F16_CEC2005 F18_CEC2005 F18_CEC2005 F19_CEC2005 F19_CEC2005 F19_CEC2005 F10_CEC205 F10_CEC25 F10_CEC25 F10_CEC25 F10_CEC25 F10_CEC25 F10_CEC	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536 0.00000 0.0000 0.0000 0.0000 0.0000 0.0000 0.00000 0.00000 0.00000 0.00000 0.00000 0.00000 0.000000	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.1006 0.0331 0.1088 0.0000 1.0000 0.5209 0.0997 0.0478 0.3519	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000 0.7244 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000 0.8301 0.9453 0.6007 1.0000 0.1382 0.6132 1.0000 1.0000 1.0000 1.0000 0.6132 1.0000 1.0000 1.0000 1.0000 0.6132 1.0000 1	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733 0.3127 0.1872 0.0966 0.1485 0.1259 0.1484	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000 0.0000 0.0000 0.1340 0.1099 0.1335 0.4907 0.1073 0.1490	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.07155 0.4866 0.7155 0.4862 0.9866 0.1252 0.9866 0.1252 0.9867 0.4474 0.1328 0.4933 0.4956 0.1364 0.1562 0.0975 0.1075	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0476 0.1094 0.0000 0.0945 0.0856 0.4266 0.0061 0.0425 0.8700 0.0588 0.4022 0.0744 0.0111 0.0089
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F6_CEC2005 F7_CEC2005 F7_CEC2005 F9_CEC2005 F10_CEC2005 F10_CEC2005 F12_CEC2005 F14_CEC2005 F14_CEC2005 F14_CEC2005 F16_CEC2005 F16_CEC2005 F19_CEC2005 F19_CEC2005 F19_CEC2005 F19_CEC2005 F10_CEC205 F10_CEC205 F10_CEC205 F10_CEC205 F10_CEC205 F10_CEC205 F10_CEC205 F10_CEC205 F10_C	GSA ave 0.4318 0.2614 0.8724 0.8724 0.8725 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.0000 0.0516 1.0000 0.0000 0.00516 1.0000 0.0000 0.0055 0.0306 0.023 0.056 0.2677	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.0331 0.1088 0.0000 1.0000 0.5209 0.0997 0.0478 0.3519 1	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.7013 1.0000 0.7244 1.0000 0.7244 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000 0.8301 0.9453 0.6007 1.0000 1.0000 1.0000 0.1528	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733 0.3127 0.1872 0.0966 0.1485 0.1259 0.1484 0.6712	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000 0.0000 0.0000 0.1340 0.1335 0.4907 0.1073 0.1490 0.0793	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.7155 0.4252 0.9867 0.4252 0.9867 0.4474 0.1328 0.4933 0.4956 0.1364 0.1300 0.1162 0.0975 0.1075 0.1075 0.1075	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476 0.1094 0.0000 0.0945 0.0856 0.4266 0.0425 0.8700 0.0425 0.8700 0.0588 0.4022 0.0744 0.0111 0.0089 0.0000
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F5_CEC2005 F6_CEC2005 F9_CEC2005 F9_CEC2005 F10_CEC2005 F10_CEC2005 F11_CEC2005 F13_CEC2005 F14_CEC2005 F16_CEC2005 F16_CEC2005 F18_CEC2005 F18_CEC2005 F19_CEC2005 F20_CEC2005 F21_CEC2005 F22_CEC2005	GSA ave 0.4318 0.2614 0.8724 0.8724 0.8725 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.0000 0.0516 1.0000 0.00000 0.0000 0.00000 0.00000 0.00000 0.00000 0.00000 0.000000	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.1006 0.0331 0.1088 0.0000 1.0000 0.5209 0.0997 0.0478 0.3519 1 0.4313	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000 0.7244 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.0382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000 0.8301 0.9453 0.6007 1.0000 1.0000 1.0000 1.0000 0.1528 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733 0.3127 0.1872 0.0966 0.1485 0.1259 0.1484 0.6712 0.2646	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000 0.0000 0.0000 0.1340 0.1099 0.1335 0.4907 0.1073 0.1490 0.0793 0.2646	FA ave 0.1076 0.1705 0.0620 0.0661 0.1075 0.9866 0.7155 0.4252 0.9867 0.4253 0.4933 0.4956 0.1328 0.4956 0.162 0.9755 0.162 0.975 0.162 0.975 0.1075	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0067 0.0476 0.1094 0.00945 0.0856 0.4266 0.0061 0.0425 0.8700 0.0588 0.4022 0.0744 0.0111 0.0089 0.0000 0.0018
F F1_CEC2005 F2_CEC2005 F3_CEC2005 F4_CEC2005 F5_CEC2005 F6_CEC2005 F8_CEC2005 F9_CEC2005 F10_CEC2005 F11_CEC2005 F12_CEC2005 F14_CEC2005 F16_CEC2005 F16_CEC2005 F18_CEC2005 F19_CEC2005 F19_CEC2005 F20_CEC2005 F21_CEC2005 F22_CEC2005 F23_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8724 0.9108 0.2817 0.4704 0.2536 0.0000 0.0516 1.0000 0.0055 0.0306 0.023 0.056 0.2677 0.1514 0.9926	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.1006 0.0331 0.1088 0.0000 1.0000 0.5209 0.0997 0.0478 0.3519 1 0.4313 0.0000	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000 0.7244 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 0.4327	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733 0.3127 0.1872 0.0966 0.1485 0.1259 0.1484 0.6712 0.2646 0.3589	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.1335 0.4907 0.1073 0.1490 0.0793 0.2646 0.5613	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0104 0.0725 0.9866 0.4252 0.9867 0.4474 0.1328 0.4933 0.4956 0.1364 0.1360 0.162 0.0975 0.162 0.0975 0.5613 0.2352 0.0206	std 0.1120 0.0191 0.0640 0.0007 0.0476 0.0476 0.1094 0.0000 0.0945 0.0856 0.4266 0.0061 0.0425 0.8700 0.0588 0.4022 0.0744 0.0111 0.0089 0.0000 0.0618 0.4696
F F1_CEC2005 F2_CEC2005 F4_CEC2005 F5_CEC2005 F6_CEC2005 F6_CEC2005 F8_CEC2005 F8_CEC2005 F10_CEC2005 F11_CEC2005 F12_CEC2005 F14_CEC2005 F15_CEC2005 F16_CEC2005 F16_CEC2005 F18_CEC2005 F19_CEC2005 F19_CEC2005 F20_CEC2005 F21_CEC2005 F21_CEC2005 F22_CEC2005 F24_CEC2005 F24_CEC2005 F24_CEC2005	GSA ave 0.4318 0.2614 0.1419 0.8724 0.8225 0.1270 0.9108 0.2817 0.4704 0.2536 0.0000 0.00516 1.0000 0.0000 0.00516 1.0000 0.0000 0.0055 0.0306 0.023 0.056 0.2677 0.1514 0.9926 0.5784	std 0.5622 0.0257 0.1081 0.4499 0.0534 0.0762 0.5975 1.0000 1 0.7690 0.4116 0.1006 0.0331 0.1088 0.0000 1.0000 0.5209 0.0997 0.0478 0.3519 1 0.4313 0.0000 0.0000	BA ave 1.0000 1.0000 1.0000 1.0000 1.0000 0.0000 0.7013 1.0000 0.9638 0.1440 1.0000 0.7244 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000	std 0.8805 1.0000 1.0000 1.0000 1.0000 0.1382 0.3595 0.8251 0.2210 0.6132 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000 1.0000	FPA ave 0.1184 0.0615 0.0000 0.0651 0.2707 0.0149 0.0708 0.9719 0.6643 0.9719 0.6643 0.5307 0.7825 0.2963 0.0824 0.4733 0.3127 0.0866 0.1485 0.1259 0.1484 0.6712 0.2646 0.3589 0.3426	std 0.3697 0.0320 0.0000 0.0685 0.3970 0.0323 0.1859 0.0924 0.3360 0.9560 0.0509 0.0000 0.0024 0.0509 0.0000 0.0000 0.0000 0.0024 0.0509 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0024 0.0000 0.01335 0.4907 0.0173 0.1490 0.0793 0.2646 0.5613 0.1616	FA ave 0.1076 0.1705 0.0620 0.0661 0.1041 0.0725 0.9866 0.7155 0.4252 0.9866 0.4474 0.1328 0.4933 0.4956 0.162 0.0975 0.162 0.0975 0.162 0.0254	std 0.1120 0.0191 0.0640 0.0000 0.0675 0.0476 0.1094 0.0945 0.0856 0.4266 0.0061 0.0425 0.8700 0.0588 0.4022 0.0744 0.0111 0.0889 0.0000 0.0618 0.4696 0.1246

Variable range
$$0.01 \le x_1, x_2, x_3, x_4, x_5 \le 100,$$

Fig. 12 shows that the cantilever beam is built using five, hollow, square-section, box girders, and the lengths of those girders are the design parameters for this problem. There is also one constraint for this problem. The GOA algorithm with 20 search agents and a maximum of 650 iterations is employed to determine the optimum for this problem. The results are presented and compared to ALO, MMA, GCA_I, GCA-II, CS, and SOS for verification in Table 9.

The results in Table 9 show that GOA finds the second best optimal weight. However, this algorithm provides the lowest number maximum function evaluation.

4.3. 52-bar truss design

In this problem, the objective is to minimise the weight of a truss with 52 bars and 20 nodes. As shown in Fig. 13, four of the nodes are fixed and the bars are classified in 12 groups as follows, which are the main parameters to be optimised for this problem:

- *Group* 1: *A*₁, *A*₂, *A*₃, *A*₄
- Group 2: A₅, A₆, A₇, A₈, A₉, A₁₀
- Group 3: A₁₁, A₁₂, A₁₃
- Group 4: A₁₄, A₁₅, A₁₆, A₁₇
- Group 5: A₁₈, A₁₉, A₂₀, A₂₁, A₂₂, A₂₃
- Group 6: A₂₄, A₂₅, A₂₆
- Group 7: A₂₇, A₂₈, A₂₉, A₃₀

Table 7						
P-values	obtained	from	the	Wilcoxon	ranksum	test.

TP	GOA	PSO	GA	DE	GSA	BA	FPA	FA
F1_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079
F2_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0159	0.0079
F3_CEC2005	0.2222	0.0079	0.0079	0.0079	0.0079	0.0079	1.0000	0.0079
F4_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0952	0.0317
F5_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079
F6_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079
F7_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079
F8_CEC2005	0.0079	0.0079	0.0079	0.0079	0.0079	1.0000	0.0079	0.0079
F9_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079
F10_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079
F11_CEC2005	0.0079	0.0079	0.0079	0.0079	N/A	0.0079	0.0079	0.0079
F12_CEC2005	0.0556	0.0079	0.0079	0.0079	N/A	0.0556	0.0079	0.0079
F13_CEC2005	N/A	0.0079	0.0079	0.0079	0.0159	0.0079	0.0079	0.0079
F14_CEC2005	0.6905	N/A	0.0556	0.0952	0.0079	0.0317	0.1508	0.1508
F15_CEC2005	0.0079	0.0079	0.0079	0.0079	N/A	0.0079	0.0079	0.0079
F16_CEC2005	0.2222	0.0952	0.0079	0.6905	N/A	0.0079	0.6905	0.6905
F17_CEC2005	N/A	0.1508	0.1508	0.8413	0.3095	0.0079	1.0000	0.4206
F18_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079
F19_CEC2005	N/A	0.0079	0.0079	0.0079	0.0159	0.0079	0.0079	0.0079
F20_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079
F21_CEC2005	N/A	0.0079	0.0079	0.0079	0.0952	0.0079	0.0079	0.0079
F22_CEC2005	N/A	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079	0.0079
F23_CEC2005	0.0317	0.0079	0.0079	0.0079	N/A	0.0079	0.0079	0.0079
F24_CEC2005	0.4206	0.0079	0.0079	0.0079	N/A	0.0079	0.0079	0.0079
F25_CEC2005	0.0079	0.0079	0.0079	0.0079	N/A	0.0079	0.0079	0.4206



Fig. 11. Three-bar truss design problem.

- Group 8: A₃₁, A₃₂, A₃₃, A₃₄, A₃₅, A₃₆
- Group 9: A₃₇, A₃₈, A₃₉
- Group 10: A₄₀, A₄₁, A₄₂, A₄₃
- Group 11: A₄₄, A₄₅, A₄₆, A₄₇, A₄₈, A₄₉
- Group 12: A₅₀, A₅₁, A₅₂

The following list presents other parameters involved in this problem:

• $\rho = 7860.0 \ kg/m^3$

- *E*=2.07*e*5 *MPa*
- Stress limitation = 180 MPa
- Maximum stress = 179.7652 MPa
- Design variabe set are chosen from Table 18

• $P_k = 100 \ kN$, $P_v = 200 \ kN$

This is a discrete problem, in which the values for the 12 parameters should be chosen from the 64 values available in Table 10. To make GOA discrete, we simply round the search agents to the nearest integer. This problem is solved using 30 search agents and 500 iterations, and the results are presented in Table 11. Inspecting the results of this table, it is evident that the GOA finds the best optimal value for this problem with the least number of function evaluations. This highlights the performance of GOA in solving real problems with more variables.

These results clearly demonstrate the merits of the GOA algorithm in solving real problems with unknown search spaces. The success of the GOA algorithm is due to several reasons. The exploration ability of GOA is high in the initial steps of optimisation, which is due to the large repulsion rate between grasshoppers. This assists GOA to explore the search space broadly and discover its promising regions. Then exploitation is high in the last steps of optimisation, which is due to the larger attraction forces between the grasshoppers. This behaviour causes local search and

Table 8		
Comparison results of th	e three-bar truss	design problem.

Algorithm	Optimal values for variables		Optimal weight	Max. Eval.
	x ₁	<i>x</i> ₂		
GOA	0.788897555578973	0.407619570115153	263.895881496069	13,000
ALO [57]	0.788662816000317	0.408283133832901	263.8958434	14,000
DEDS [58]	0.78867513	0.40824828	263.8958434	15,000
PSO-DE [59]	0.7886751	0.4082482	263.8958433	17,600
MBA [56]	0.7885650	0.4085597	263.8958522	20,000
Ray and Sain [60]	0.795	0.395	264.3	N/A
Tsa [61]	0.788	0.408	263.68	N/A
CS [55]	0.78867	0.40902	263.9716	15,000



Fig. 12. Cantilever beam design problem.

Table 9						
Comparison	results	for	cantilever	design	problem.	

Algorithm	Optimal values for variables					Optimal	Max.
	<i>x</i> ₁	<i>x</i> ₂	<i>x</i> ₃	<i>x</i> ₄	<i>x</i> ₅	weight	Eval.
GOA	6.011674	5.31297	4.48307	3.50279	2.16333	1.33996	13,000
ALO [57]	6.01812	5.31142	4.48836	3.49751	2.158329	1.33995	14,000
MMA [62]	6.0100	5.3000	4.4900	3.4900	2.1500	1.3400	N/A
GCA_I [62]	6.0100	5.3000	4.4900	3.4900	2.1500	1.3400	N/A
GCA_II [62]	6.0100	5.3000	4.4900	3.4900	2.1500	1.3400	N/A
CS [55]	6.0089	5.3049	4.5023	3.5077	2.1504	1.33999	2500
SOS [63]	6.01878	5.30344	4.49587	3.49896	2.15564	1.33996	15,000

improving the accuracy of the solution(s) obtained in the exploration phase. The algorithm smoothly balances exploration and exploitation, initially emphasising local optima avoidance and then convergence. This behaviour is due to the proposal of the adaptive comfort zone coefficient. The gradual decrementing of this component brings the grasshopper closer to the target proportional to the number of iterations. Finally, the proposed target chasing mechanism requires GOA to save the best solution obtained so far as the target and drive the grasshoppers towards it with the hope of improving its accuracy or finding a better one in the search space.

Considering the simulations, results, discussion, and analyses of this paper, we believe that GOA is able to solve many optimisation problems effectively. GOA considers a given optimisation problem as a black box, so it does not need gradient information of the search space. Therefore, this algorithm can be applied to any optimisation problem in different fields subject to proper problem formulation.

5. Conclusion

This work proposed an optimisation algorithm called the Grasshopper Optimisation Algorithm. The proposed algorithm mathematically modelled and mimicked the swarming behaviour of grasshoppers in nature for solving optimisation problems. A mathematical model was proposed to simulate repulsion and attraction forces between the grasshoppers. Repulsion forces allow grasshoppers to explore the search space, whereas attraction forces encouraged them to exploit promising regions. To balance between exploration and exploitation, GOA was equipped with a coefficient that adaptively decreases the comfort zone of the grasshoppers. Finally, the best solution obtained so far by the swarm was considered as a target to be chased and improved by the grasshoppers.

In order to benchmark the performance of the proposed algorithm, a series of tests was conducted. Firstly, a set of 2D test functions was solved by the GOA to observe its performance qualitatively. This experiment and relevant discussions support the following conclusions:

• Grasshoppers effectively discover the promising regions of a given search space.



Fig. 13. Structure of a 52-bar truss.

- Grasshoppers face abrupt, large-scale changes in the initial steps of optimisation, which assist them to search globally.
- Grasshoppers tend to move locally in the final steps of optimisation, which allows them to exploit the search space.

Table 10							
Available cross-section	areas of th	e AISC no	orm (valid	values	for th	e paramete	ers).

		2		. 2	2
No.	ın.²	mm²	No.	ın.²	mm ²
1	0.111	71.613	33	3.84	2477.414
2	0.141	90.968	34	3.87	2496.769
3	0.196	126.451	35	3.88	2503.221
4	0.25	161.29	36	4.18	2696.769
5	0.307	198.064	37	4.22	2722.575
6	0.391	252.258	38	4.49	2896.768
7	0.442	285.161	39	4.59	2961.284
8	0.563	363.225	40	4.8	3096.768
9	0.602	388.386	41	4.97	3206.445
10	0.766	494.193	42	5.12	3303.219
11	0.785	506.451	43	5.74	3703.218
12	0.994	641.289	44	7.22	4658.055
13	1	645.16	45	7.97	5141.925
14	1.228	792.256	46	8.53	5503.215
15	1.266	816.773	47	9.3	5999.988
16	1.457	939.998	48	10.85	6999.986
17	1.563	1008.385	49	11.5	7419.34
18	1.62	1045.159	50	13.5	8709.66
19	1.8	1161.288	51	13.9	8967.724
20	1.99	1283.868	52	14.2	9161.272
21	2.13	1374.191	53	15.5	9999.98
22	2.38	1535.481	54	16	10,322.56
23	2.62	1690.319	55	16.9	10,903.2
24	2.63	1696.771	56	18.8	12,129.01
25	2.88	1858.061	57	19.9	12,838.68
26	2.93	1890.319	58	22	14,193.52
27	3.09	1993.544	59	22.9	14,774.16
28	3.13	2019.351	60	24.5	15,806.42
29	3.38	2180.641	61	26.5	17,096.74
30	3.47	2238.705	62	28	18,064.48
31	3.55	2290.318	63	30	19,354.8
32	3.63	2341.931	64	33.5	21,612.86

- The varying comfort zone coefficient requires grasshoppers to gradually balance exploration and exploitation, which helps GOA not to become trapped in local optima and find an accurate approximation of the global optimum.
- The GOA algorithm enhances the average fitness of grasshoppers, which shows that this algorithm is able to effectively improve the initial random population of grasshoppers.
- The fitness of target is improved over the course of iterations, which shows that the approximation of the global optimum becomes more accurate proportional to the number of iterations.

After the first experiment, four sets of challenging test functions were employed. The test functions were unimodal, multimodal, composite, and CEC2005. The GOA algorithm managed to outperform several algorithms in the literature. The findings and discussions of the second experiment support the following conclusions:

- Exploitation of the GOA is satisfactory on problems involving unimodal test functions.
- Exploration of the GOA is intrinsically high for multi-modal test functions.
- GOA properly balances exploration and exploitation when solving challenging problems involving composite test functions.
- GOA has the potential to significantly outperform several current algorithms when solving a range of current or new optimisation problems.

The last experiment was performed on three real problems in the field of structural design. All the problems were successfully solved, which demonstrates the practical merits of the proposed algorithm. From the results, findings, and discussions of the real applications, the following conclusions can be drawn:

- GOA is able to improve the initial random population for a real problem.
- The target is improved over the course of iterations, so the approximation of the global optimum become more accurate proportional to the number of iterations.
- GOA is able to solve real problems with unknown search spaces.

GOA is only able to solve single-objective problems with contentious variables. For future work, binary and multi-objective versions of this algorithm may be developed to solve discrete and multi-objective problems. The comfort zone parameter is an important coefficient in GOA, so it is worth investigating the impacts of different comfort zone functions on the performance of the algorithm. Solving optimisation problems in different fields could also be a valuable contribution. Tuning the main controlling parameters of GOA may also be beneficial.

Appendix

Tables 12-15

Ta	hle	11	

Comparison of GOA	optimisation	results with	literature for	the 52-bar	truss de	esign p	problem
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Variables (mm ²)	PSO [64]	PSOPC [64]	HPSO [64]	DHPSACO [65]	MBA [66]	SOS [63]	GOA
A1 - A4	4658.055	5999.988	4658.055	4658.055	4658.055	4658.055	4658.055
A5 - A10	1374.19	1008.38	1161.288	1161.288	1161.288	1161.288	1161.288
A11 - A13	1858.06	2696.77	363.225	494.193	494.193	494.193	494.193
A14 - A17	3206.44	3206.44	3303.219	3303.219	3303.219	3303.219	3303.219
A18 - A23	1283.87	1161.29	940	1008.385	940	940	940
A24 - A26	252.26	729.03	494.193	285.161	494.193	494.193	494.193
A27 - A30	3303.22	2238.71	2238.705	2290.318	2238.705	2238.705	2238.705
A31 - A36	1045.16	1008.38	1008.385	1008.385	1008.385	1008.385	1008.385
A37 - A39	126.45	494.19	388.386	388.386	494.193	494.193	494.193
A40 - A43	2341.93	1283.87	1283.868	1283.868	1283.868	1283.868	1283.868
A44 - A49	1008.38	1161.29	1161.288	1161.288	1161.288	1161.288	1161.288
A50 - A52	1045.16	494.19	792.256	506.451	494.193	494.193	494.193
Optimal weight (kg)	2230.16	2146.63	1905.495	1904.83	1902.605	1902.605	1902.605
No. of analyses	150,000	150,000	5300	11,100	5450	2350	2300

Table 12 Unimodal benchmark functions.

Function	Dim	Range	f _{min}
$f_1(x) = \sum_{i=1}^n x_i^2 f_2(x) = \sum_{i=1}^n x_i + \prod_{i=1}^n x_i $	30	[-100,100]	0
	30	[-10,10]	0
$f_3(x) = \sum_{i=1}^{i} \left(\sum_{j=1}^{i} x_j\right)^2 f_4(x) = \max_i x\{ x_i , \ 1 \le i \le n\}$	30	[-100,100]	0
	30	[-100,100]	0
$\begin{split} f_5(x) &= \sum_{i=1}^{n-1} \left[100(x_{i+1} - x_i^2)^2 + (x_i - 1)^2 \right] \\ f_6(x) &= \sum_{i=1}^n \left(x_i + 0.5 \right)^2 \\ f_7(x) &= \sum_{i=1}^n ix_i^4 + random[0, 1) \end{split}$	30	[-30,30]	0
	30	[-100,100]	0
	30	[-1.28,1.28]	0

 Table 13

 Multimodal benchmark functions.

Function	Dim	Range	f _{min}
$F_8(x) = \sum_{i=1}^n -x_i \sin(\sqrt{ x_i })$	30	[-500,500]	-418.9829 × Dim
$F_{9}(x) = \sum_{i=1}^{n} [x_{i}^{2} - 10\cos(2\pi x_{i}) + 10]$	30	[-5.12,5.12]	0
$F_{10}(x) = -20exp(-0.2\sqrt{\frac{1}{n}}\sum_{i=1}^{n}x_i^2) - exp(\frac{1}{n}\sum_{i=1}^{n}\cos(2\pi x_i)) + 20 + e$	30	[-32,32]	0
$F_{11}(\mathbf{x}) = \frac{1}{4000} \sum_{i=1}^{n} x_i^2 - \prod_{i=1}^{n} \cos(\frac{x_i}{\sqrt{i}}) + 1$	30	[-600,600]	0
$F_{12}(x) = \frac{\pi}{n} \{ 10\sin(\pi y_1) + \sum_{i=1}^{n-1} (y_i - 1)^2 [1 + 10\sin^2(\pi y_{i+1})] + (y_n - 1)^2 \} + \sum_{i=1}^n u(x_i, 10, 100, 4) + \sum_{i=1}^n u(x_i, 10, 100, 4) \}$			
$y_i = 1 + \frac{x_i + 1}{4}$			
$\int k(x_i - a)^m x_i > a$			
$u(x_i, a, k, m) = \begin{cases} 0 - a < x_i < a \end{cases}$	30	[-50,50]	0
$k(-x_i-a)^m x_i < -a$			
$F_{13}(x) = 0.1\{\sin^2(3\pi x_1) + \sum_{i=1}^n (x_i - 1)^2 [1 + \sin^2(3\pi x_i + 1)] + (x_n - 1)^2 [1 + \sin^2(2\pi x_n)]\} + \sum_{i=1}^n u(x_i, 5, 100, 4)$	30	[-50,50]	0

Table 14Composite benchmark functions.

Function	Dim	Range	\mathbf{f}_{\min}
F ₁₄ (CF1):	30	[-5,5]	0
$f_1, f_2, f_3,, f_{10} =$ Sphere Function			
$[6_1, 6_2, 6_3, \dots, 6_{10}] = [1, 1, 1, \dots, 1]$			
$[\lambda_1, \lambda_2, \lambda_3, \lambda_{10}] = [5/100, 5/100,, 5/100]$			
F ₁₅ (CF2):	30	[-5,5]	0
$f_1, f_2, f_3, \ldots, f_{10} = Griewank's Function$			
$[6_1, 6_2, 6_3, \dots, 6_{10}] = [1, 1, 1,, 1]$			
$[\lambda_1, \lambda_2, \lambda_3,, \lambda_{10}] = [5/100, 5/100, 5/100,, 5/100]$			
F ₁₆ (CF3):	30	[-5,5]	0
$f_1, f_2, f_3, \ldots, f_{10} = $ Griewank's Function			
$[6_1, 6_2, 6_3, \dots, 6_{10}] = [1, 1, 1, \dots, 1]$			
$[\lambda_1, \lambda_2, \lambda_3,, \lambda_{10}] = [1, 1, 1,, 1]$			
f ₁₇ (CF4):	30	[-5,5]	0
$f_1, f_2 = \text{Ackley'sFunction}$			
$f_3, f_4 = \text{Rastrigin's Function}$			
f_5, f_6 = Weierstrass Function			
$f_7, f_8 =$ Griewank's Function			
$f_9, f_{10} = $ Sphere Function			
$[6_1, 6_2, 6_3, \dots, 6_{10}] = [1, 1, 1, \dots, 1]$			
$[\lambda_1, \lambda_2, \lambda_3,, \lambda_{10}] = [5/32, 5/32, 1, 1, 5/0.5, 5/100, 5/100, 5/100, 5/100]$	20	(
I ₁₈ (CF5):	30	[-5,5]	0
$J_1, J_2 = \text{Rastrigue's Function}$			
f_3, f_4 = Weierstrass Function			
J ₅ , J ₆ = Griewank's Function			
$J_7, J_8 = Ackney Srunction$			
$f_{9}, f_{10} = $ Sphere Function			
$[0_1, 0_2, 0_3, \dots, 0_10] = [1, 1, 1, \dots, 1]$			
$[x_1, x_2, x_3,, x_{10}] = [1/3, 1/3, 5/0.3, 5/0.3, 5/100, 5/100, 5/52, 5/52, 5/100, 5/100]$	20	[55]	0
f_{19} (CPO).	30	[-3,5]	0
$j_1, j_2 = \text{Rastright's Function}$			
13, 14 = Weielstrass Function			
$J_5, J_6 = \text{Chevark STutcton}$			
J, J8 = Acted Stateton			
$f_{9}, f_{10} = 5$ and $f_{10} = 6$ and $f_{10} = 6$			
$[\lambda_1, \lambda_2, \lambda_3, \dots, \lambda_{10}] = [0.1, 0.2, 0.3, 0.1, 0.3, 0.0, 0.7, 0.0, 0.5, 1]$			
0.6* 5/100, 0.7*5/32, 0.8* 5/32, 0.9*5/100, 1*5/100]			

Table 15

CEC2005 test functions.

Function	Dim	Range	f _{min}
F1_CEC2005: Shifted Sphere Function	30	[-100,100]	-450
F2_CEC2005: Shifted Schwefel's Problem	30	[-100,100]	-450
F3_CEC2005:Shifted Rotated High Conditioned Elliptic Function	30	[-100,100]	-450
F4_CEC2005: Shifted Schwefel's Problem with Noise in Fitness	30	[-100,100]	-450
F5_CEC2005: Schwefel's Problem with Global Optimum on Bounds	30	[-100,100]	-310
F6_CEC2005: Shifted Rosenbrock's Function	30	[-100,100]	390
F7_CEC2005: Shifted Rotated Griewank's Function without Bounds	30	[-600, 600]	-180
F8_CEC2005: Shifted Rotated Ackley's Function with Global Optimum on Bounds	30	[-32,32]	-140
F9_CEC2005: Shifted Rastrigin's Function	30	[-5,5]	-330
F10_CEC2005: Shifted Rotated Rastrigin's Function	30	[-5,5]	-330
F11_CEC2005: Shifted Rotated Weierstrass Function	30	[-0.5,0.5]	90
F12_CEC2005: Schwefel's Problem	30	[-100,100]	-460
F13_CEC2005: Expanded Extended Griewank's plus Rosenbrock's Function (F8F2)	30	[-3,1]	-130
F14_CEC2005: Expanded Rotated Extended Scaffe's F6	30	[-100,100]	-300
F15_CEC2005: Hybrid Composition Function 1	30	[-5,5]	120
F16_CEC2005: Rotated Hybrid Composition Function 1	30	[-5,5]	120
F17_CEC2005: Rotated Hybrid Composition Function 1 with Noise in Fitness	30	[-5,5]	120
F18_CEC2005: Rotated Hybrid Composition Function 2	30	[-5,5]	10
F19_CEC2005: Rotated Hybrid Composition Function 2 with a Narrow Basin for the Global Optimum	30	[-5,5]	10
F20_CEC2005: Rotated Hybrid Composition Function 2 with the Global Optimum on the Bounds	30	[-5,5]	10
F21_CEC2005: Rotated Hybrid Composition Function 3	30	[-5,5]	360
F22_CEC2005: Rotated Hybrid Composition Function 3 with High Condition Number Matrix	30	[-5,5]	360
F23_CEC2005: Non-Continuous Rotated Hybrid Composition Function 3	30	[-5,5]	360
F24_CEC2005: Rotated Hybrid Composition Function 4	30	[-5,5]	260
F25_CEC2005: Rotated Hybrid Composition Function 4 without Bounds	30	[-5,5]	260

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